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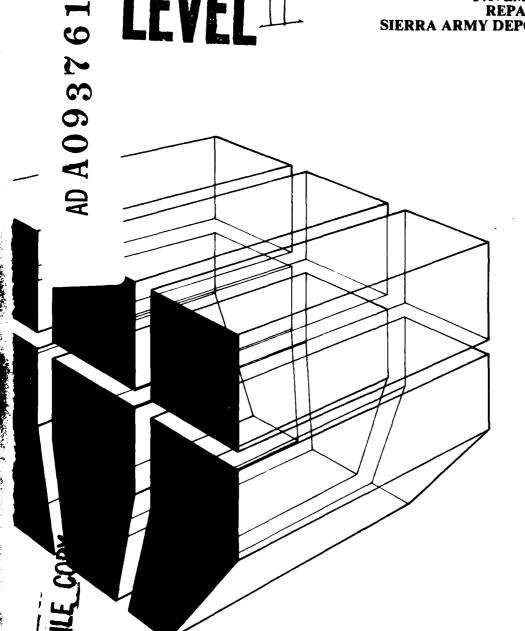
construction engineering research laboratory



TECHNICAL REPORT M-283 November 1980

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PAVEMENT EVALUATION AND REPAIR RECOMMENDATION SIERRA ARMY DEPOT, AMEDEE AIR STRIP



M. Y. Shahin

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FOREWORD

This research was conducted for the Sierra Army Depot, Herlong, CA, under IAO PR 9-80 by the U.S. Army Construction Engineering Research Laboratory, Champaign, Il.. The Sierra Army Depot Project Monitor was Mr. Michael P. Balerviez. The CERL Principal Investigator was Dr. M. Y. Shahin.

The following people are acknowledged for their participation in surveying the runway: Mssrs. Mike Flaherty and Jim West, DARCOM; Mssrs. Mike Balerviez, Ray McMillan, and David Wickward, Sierra Army Depot. Dr. R. Quattrone is Chief of CERL-EM, COL Louis J. Circeo is Commander and Director of CERL, and DR. L. R. Shaffer is Technical Director.

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PAVEMENT EVALUATION AND REPAIR RECOMMENDATION SIERRA ARMY DEPOT, AMEDEE AIR STRIP

1 INTRODUCTION

Background and Description of Existing Facility

Amedee Air Strip is located at Sierra Army Depot, Herlong, California, east of Honey Lake and 50 air miles northwest of Reno at an elevation of 4000 feet.

A layout of the runway is shown in Figure 1. The runway surface is a 3-inch layer of asphalt concrete (AC) which, with the 4-inch crushed base below, was added in 1969. The original runway was constructed in 1943, and the 1969 overlay was emplaced because of considerable cracking in the original surface. The 4-inch crushed stone base was used to eliminate (or minimize) reflection cracking in the new surface.

Since its overlay in 1969, the runway has had a history of problems, and experts from Office, Chief of Engineers (OCE), Sacramento District Engineer Office, Defense Acquisition and Readiness Command (DARCOM), Depot Systems Command (DESCOM), and the U.S. Army Corps of Engineers Waterways Experiment Station (WES) have made on-site evaluations of its condition. These evaluations showed that although the runway appears to be structurally sound, the pavement has oxidized, causing surface cracks. It was also concluded that the cracks are not reflective cracks from the old pavement constructed in 1943.

The cost of maintaining the runway from 1969 to 1979 1.38 been \$320,550, and is increasing rapidly.

Objective

The objective of this study was to determine the optimum maintenance and repair alternative for Amedee Airfield.

2 INVESTIGATION

Runway Condition Survey

On 13-15 November 1979 a team from DAR-COM, the Sierra Army Depot, and CERL performed

a condition survey of the runway using the Pavement Condition Index (PCI) procedure developed by CERI, and implemented by the U.S. Air Force worldwide. The runway was divided into distinct features based on structural composition and traffic distribution as shown in Figure 2. Each feature was divided into sample units for inspection. The number and location of sample units inspected were determined as shown in Figure 3. A PCI inspection was performed on all the runway features, and the results are shown in parentheses in Figure 2. Figure 4 shows a plot of the PCI for the individual sample units of each feature, and Appendix A provides a complete PCI computer output.

The results of the PCI survey (Appendix A) showed that the distress is mostly linear cracking. Figure 5 is a photograph of the runway surface. Table 1 summarizes the quantities and severities of linear cracking found in each feature and Table 2 summarizes the other distresses that were found.

Field Investigation

From the PCI inspection, it was determined that the majority of the cracks are temperature related; however, whether the distress was reflective, started from the base course (i.e., bottom-up), or started from the surface (i.e., top-down) was not clear. Therefore, three cuts (approximately 1 x 2 feet) were made around the different cracks so that their cause and characteristics could be clearly determined (Figure 6). Cuts #1 and #3 were outside the traffic area, and cut #2 was in the traffic area.

Cut #1 (Figure 7) was across a 4-inch-wide crack extending through the full depth of the top AC surface. There was no evidence of the crack in the base or on the old AC surface below.

Cut #2 (Figure 8) was made across two cracks, one of medium severity and one of low severity. Both cracks were continuous throughout the full depth of the top AC surface but not through the base or old AC surface. The low-severity crack was only about 1½ feet from the medium-severity crack and both were in the wheel path, as indicated by the tire markings. These two cracks were also beginning to be connected by random cracking. It was evident that the initial temperature cracking was becoming alligator cracking because of the weakened pavement condition around the temperature cracks.

Airfield Pavement Evaluation Program, AFR 93-5 (Department of the Air Force, 1980).

Cut #3 (Figure 9) was made across the tip of a hairline crack in a nontraffic area. The cut showed that only portions of the crack had propagated to the bottom of the AC surface. In Cut #3, the surface crack was 8 inches long and only 3 inches had propagated to the bottom. Therefore it was concluded the crack had originated at the surface.

Based on the field investigation, it was concluded that the cracking in the Sierra Army Depot runway is limited to the AC surface. Furthermore, it was speculated that the temperature cracks are caused by thermal fatigue in the AC surface resulting from the high daily temperature cycling variation in this area.

Laboratory Investigations

The laboratory investigations were designed to verify field observations that the AC surface had oxidized and that the cracks were caused primarily by thermal fatigue cracking. The AC slabs obtained from the three cuts described above were forwarded to WES for testing. Appendix B provides the results of the testing.

Tests performed included: AC Marshall Stability, flow, percent voids in total mix, percent voids filled; and asphalt penetration, softening point, and viscosity. The results showed that the pavement had oxidized, as indicated by a penetration of 15 and softening point of 71.2°C (160°F) for the asphalt in the surface course.

Construction records from 1969 showed that the asphalt had an original penetration of 90 and a softening point of 49.4°C (121°F). The construction records also showed that penetration (percent of original) after the Thin Film Test was 72. The drop in penetration from 90 in 1969 to 15 in 1979 is considered high.

In addition to the above tests, the indirect tensile test was performed at four temperatures (-20, 20, 50, and 75°F) at loading rate .05 inches/minute, and at three temperatures (20, 50, and 75°F) at loading rate 2.0 inches/minute. Figure 10 is a plot of the tensile strength of the AC mix (top 1.5 inch) versus temperature.

To verify the cause of cracking, the program developed by Shahin² was used. The program predicts both low-temperature and thermal-fatigue cracking as

a function of the AC mixture properties and climatic factors. Figure 11 shows the input to the program. Figure 12 is a plot of cracking versus age as predicted from the program. As shown in the figure, there is a close agreement between the measured and predicted amounts of cracking.

Detailed analysis of the program output showed that the cracks are caused by thermal fatigue cracking (resulting from daily temperature cycling) rather than just simple low temperature. The close agreement in prediction is encouraging in that the same program can be used for future mix design and selection of optimum asphalt grade to minimize cracking. It is believed that a careful mix design and careful selection of asphalt grade can increase the pavement life by several years.

3 EVALUATION

Evaluation of Past Performance and Selection of Fessible Maintenance and Repair (M&R) Alternatives

The evaluation, which was performed according to the M&R guidelines CERL developed for the U.S. Air Force,³ was performed for feature RC3 since it is the largest feature, receives most of the traffic, and has the lowest PC1. Following is a brief discussion of the results, which are summarized in Figure 13.

- 1. The PCI of the feature is 61, which locates the feature in an M&R zone of routine, major, and overall. This is based on the guidelines shown in Figure 14, which were developed by a group of experienced Air Force engineers and subjected to considerable field testing and validation. It is to be noted that the M&R zone reflects needed M&R within 2 years of the PCI survey date.
- 2. Localized variation exists. Variation results because one sample unit has a PCI of 33 (sample unit #20), while the average PCI of the feature is 61.
- 3. The long-term rate of deterioration is high compared to other airfield AC pavements of the same

²M. Y. Shahin, "Prediction of Low-Temperature and Thermal-Fatigue Cracking in Flexible Pavements," Ph.D. Dissertation (University of Texas at Austin).

³M. Y. Shahin, Development of a Pavement Maintenance Management System, Vol VI: M&R Guidelines—Validation and Field Applications, ESL-TR-79-18 (USAF Engineering and Services Center [AFESC]).

age throughout the United States. This is illustrated in Figure 15.

4. Analysis of the load-carrying capacity showed the pavement to be structurally adequate (see Figure 16). Distress evaluation showed that 56 percent of the deduct value stem from load-associated distress (alligator cracking). However, this can be attributed to the weakened areas adjacent to cracks caused by temperature variations.

Application of the M&R Performance Standards recently developed for the U.S. Air Force* to the results of the evaluation in Figure 13 showed that most experienced maintenance engineers would consider the following M&R alternatives:

- (a) Routine,
- (b) Surface Treatment,
- (c) Thin Overlay, and
- (d) Recycling or Replacement of Surface.

The above alternatives all seemed feasible. Selection of a specific M&R alternative is a function of future performance and life-cycle costing.

Prediction of Future Performance of Selected M&R Alternatives

Five Specific M&R alternatives were analyzed:

Alternative A: Continue to seal cracks to a minimum (acceptable) PCI; then overlay with 3-inch AC at center, tapered to 1 inch at edges.

Alternative B: Seal cracks and overlay immediately with 3-inch AC at center, tapered to 1 inch at edges.

Alternative C: Replace entire surface with a new 3-inch-deep AC hot mix.

Alternative D: Recycle surface and reuse as base; then add new 3-inch AC for central 75 feet and taper to 1 inch at edges.

Alternative E: Replace central 75 feet of surface course with 3-inch AC hot mix, and continue to crack seal outside areas.

Along with each of these alternatives, it was assumed that a rejuvenating surface treatment would

be applied periodically to retard surface brittleness and thus temperature cracking.

The PCI for each alternative was predicted, using a computer program based on M&R consequence models developed for the U.S. Air Force.⁴ Appendix C provides the program output for each M&R alternative. Figure 17 is a plot of the expected PCI over time for each alternative.

Life-Cycle Costing of Selected M&R Alternatives

The life-cycle costing is determined based on initial cost, future M&R cost, and salvage value. The present-worth method was used to consider both interest and inflation rates. Figures 18 through 22 provide work summary and initial cost estimates for each alternative.

Future cracking had to be predicted in order to estimate future M&R cost. The maximum cracking expected to occur in the future, is block cracking with an average size of 10 feet x 10 feet. This translates into a total cracking length of approximately 197,050 linear feet for an area that is 150 feet x 6800 feet. The total amount of cracking currently existing is 51,053 feet. Using statistical techniques, future cracks were predicted: (see Table 3). Appendix D provides the computer output used in the prediction. Future M&R was computed on a two-year basis, assuming a repair cost of \$1.0/linear foot. Another assumption in the computation of future M&R was that cracks must be resealed every 6 years. Table 3 shows all cost calculations.

For M&R Alternative E, where only the central 75 feet would be replaced, it was essential to do the crack prediction for only the outside 75 feet. Table 4 summarizes the cracking outside the central 75 feet.

The maximum cracking expected to occur outside the central 75 feet will be in the form of block cracking having an average size of 10 feet x 10 feet, or a total

^{*}U.S. Air Force Pavement Major Command Engineers meeting held at CERI, 15-17 Jul 80.

⁴M. Y. Shahin, M. I. Darter, and T. T. Chen, Development of a Pavement Maintenance Management System, Vol VII: Maintenance and Repair Consequence Models and Management Information Requirements, FSL-TR-79-18 (AFESC, December 1979)

⁸M. Y. Shahin, M. I. Darter, and S. D. Kohn, Development of a Pavement Maintenance Management System, Vol IV: Appendices A through I. Maintenance and Repair Guidelines for Airfield Pavements, CEEDO-TR-77-44 (AFESC, 1977).

cracking length of approximately 105,325 feet. The total cracking currently existing is 18,906 feet. Using statistical techniques, future cracks were predicted (see Table 5). Appendix D provides the computer output used in the prediction. Another assumption in the computation of future M&R was that cracks must be resealed every 6 years. Table 5 shows all cost calculations.

The information in Tables 3 and 5 was used to compute future M&R costs for each alternative. (Appendix E shows the computation of future costs.) The cost information was then input to a present-worth economic analysis program. Figure 23 shows the results of the cost analysis, with ranking of alternatives based on net present cost shown at the top of the figure. Considering the amount of predictions and estimates involved, the difference in cost among the various alternatives is not large enough to allow selection of an alternative based on cost alone.

4 CONCLUSIONS AND RECOMMENDATIONS

The runway AC surface had oxidized, as shown by a measured asphalt penetration of 15 (1979) versus an original penetration of 90 (1969). Large daily temperature variations (average daily temperature range of 40°F) have caused the oxidized AC surface to crack. The amount of cracking is expected to increase at a high rate, as predicted in Table 3.

Figure 2 shows the PCI of the various runway features. The lowest PCI is 61 for feature RC3, which has been caused by further breakdown of the cracks under load.

Five feasible M&R alternatives were identified and analyzed. Figure 17 shows the performance (PCI over time) expected for each alternative. Figure 23 shows the results of life-cycle costing for each alternative. The most costly alternative (C) is only about 30 percent higher than the least costly alternative. Therefore, considering the amount of predictions and assumptions necessary to perform the life-cycle costing, the difference in net present cost among the various alternatives cannot be used as a sole indicator for

selecting the best alternative. Another factor to consider is the dollars spent per unit performance. This is computed by dividing the net present cost for each alternative by the area between the PCI (Figure 17) and the minimum acceptable PCI during the analysis period (1980 to 2000). Table 6 shows the results of these computations. Although the differences are still narrow, alternatives B, D, and E appear to be more advantageous.

Based on the overall analysis, it is recommended that alternative D be adopted; i.e., recycle surface, reusing it as base, then add new 3-inch AC for the central 75 feet, and taper it to 1 inch at the edges. Alternative D is recommended, because it offers the following unique advantages:

- 1. It is the strongest alternative structurally of great importance in case of the heavy traffic operations.
- 2. It requires the least amount of future maintenance and thus less frequent traffic interruptions.
- 3. It will eliminate the possibility of reflection cracking by recycling the surface and using it as a base.
- It provides an environmental advantage because of recycling.

It is recommended that alternative D be implemented within the coming 3 years (1981 to 1984).

It must be emphasized that any new AC mix should be carefully designed to minimize temperature cracking. Special attention should be given to the asphalt grade and specifications. Acceptance of the mix should be based on analysis similar to that described in the section on "Laboratory Investigation".

The asphalt selected should have a penetration of approximately 120 and percent penetration after the thin film oven test of 65-75. If no asphalt supplier in the area can meet the required specifications, then consideration should be given to reconstructing the runway with concrete. Concrete was not analyzed in detail, since the initial cost was estimated to be four times that of alternative D. However, if no asphalt supplier can meet the requirements to minimize cracking, then reconstruction with concrete may be economically justified based on the life cycle costs.

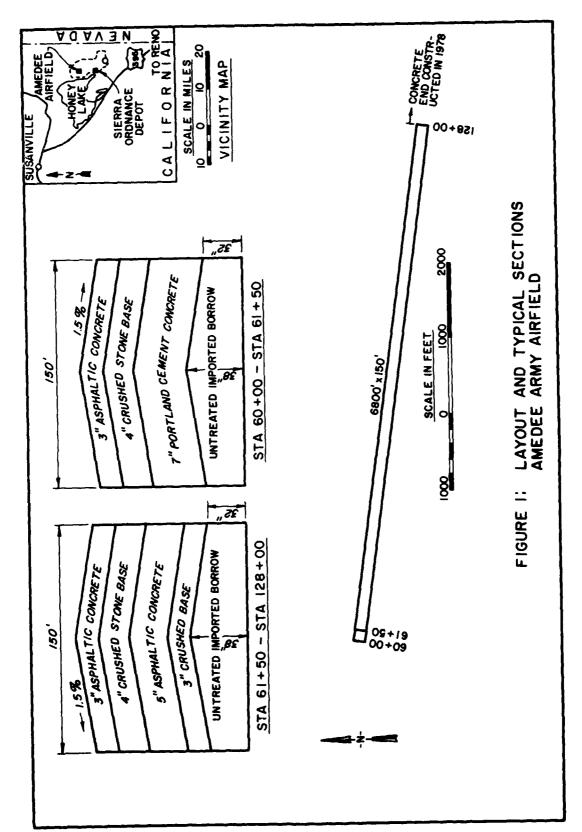


Figure 1. Layout and typical sections Amedee Army Airfield.

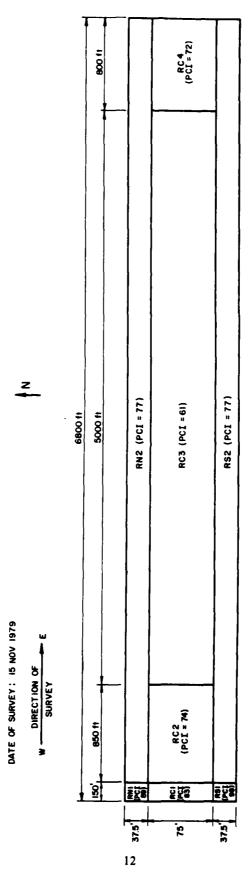


Figure 2. Division of runway into features.

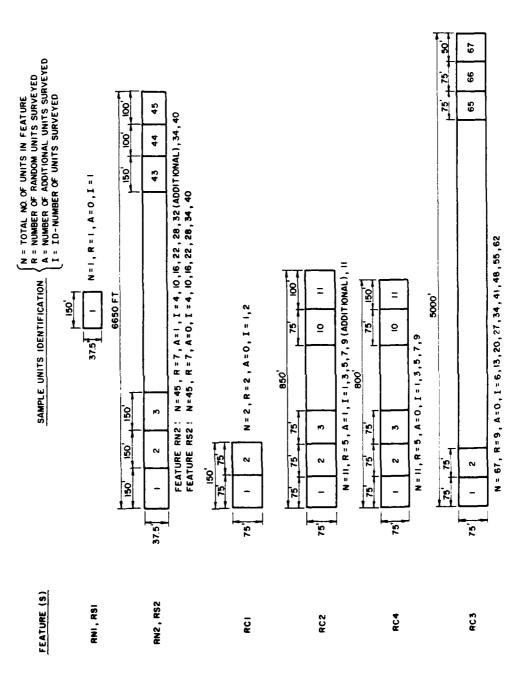


Figure 3. Division of features into sample units.

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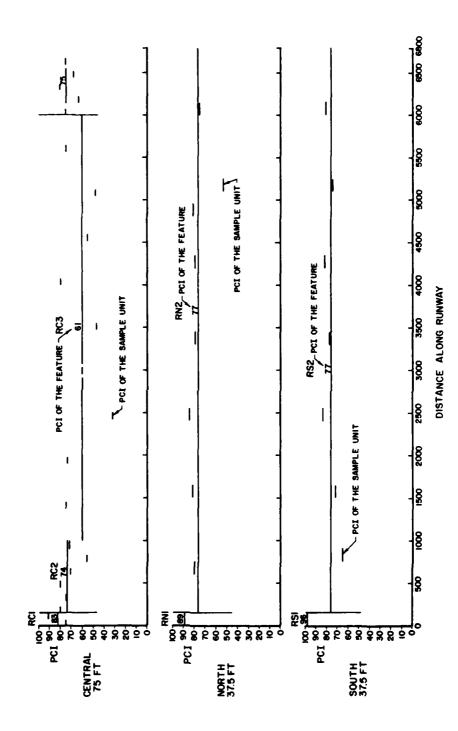


Figure 4. PCI of individual sample units for central, north, and south features.



Figure 5. Photograph of Sierra Army Depot, Amedee Runway, showing primary type of distress (temperature cracking).

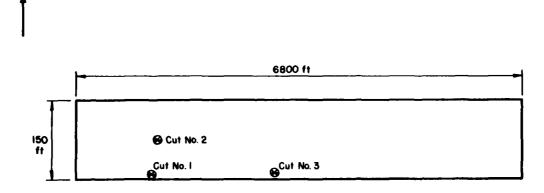
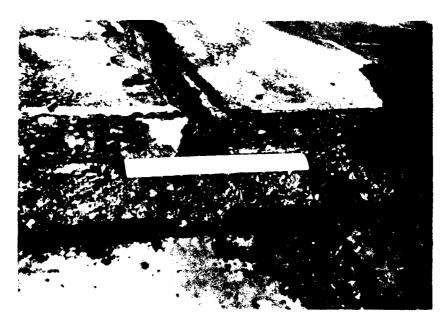


Figure 6. Runway layout showing location of cuts.

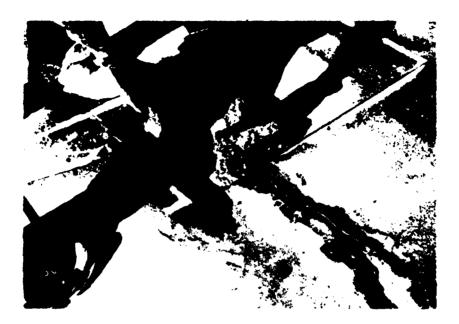


(a) Pavement before cut showing 4-inch wide crack.



(b) Pavement after cut showing crack to extend through the full depth of the top AC surface.

Figure 7. Cut #1.

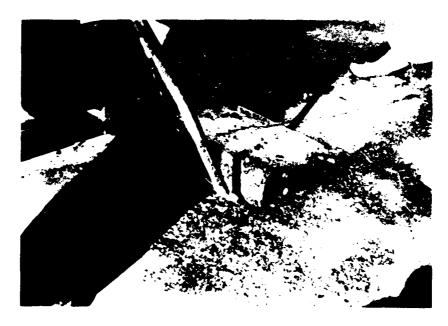


(a) Cut showing medium and low severity cracks.



(b) Bottom of pavement slab. Cracks were found to extend through the full depth of the top AC surface.

Figure 8. Cut #2.



(a) Pavement slab being carefully lifted after saw cutting.



(b) Top of slab showing 8-inch-long crack.

Figure 9. Cut #3.



(c) Bottom of slab showing 3-inch crack propagated to bottom.

Figure 9. (continued)

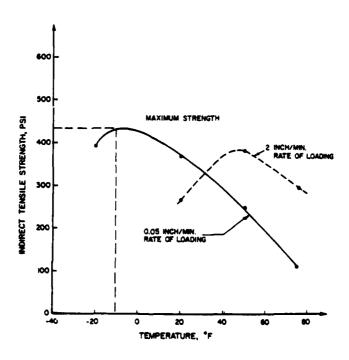


Figure 10. Tensile strength vs. temperature for AC surface top 1.5 inch.

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                FEB.
                          HAR.
                                                         JUNE
                                              MAY.
                                    10
                            AIR TEMPERATURE
                              ,DEB.F
     ANNUAL AVERAGE
                                                       47.500
     ANNUAL RANGE
                               ,DEB.F
                                                       39.600
     DAILY RANGE
                               ,DEG.F
                                                       34.500
                            FACTORS AFFECTING PAY. TEMP.
     ANNUAL AVE. SOLAR RAD. ,LANGLEYS
JULY AVE. SOLAR RAD. ,LANGLEYS
ANNUAL AVE. WIND VEL. ,HPH.
                                                      455.000
                                                      730.000
                                                        6.300
     SURFACE ABSORBTIVITY
                                                         .950
     DEPTH FOR CALCUALTION, IN.
                                                        0.000
     MIX. COMBUCTIVITY ,BTU-FT-HR-F. MIX. SPECIFIC HEAT ,BTU-LB-F.
                                                        .700
                                                         .220
                             ,LB/FT3
     HIX. DENSITY
                                                      140.200
                            ASPHALT PROPERTIES
     ORIG. PENETRATION
                               ,DMM-SSEC.
                                                          90.
     PEM. TEST TEMP. ,DEG.I
ORIG. SOFTENING POINT,DES.F
                                DEG.F
                                                          77.
                                                         121.
     THIN FILM OVEN TEST ,PCT.ORIG.PEN.
                                                       72.000
                            MIXTURE PROPERTIES
                             ,BY UT.OF AGE.
     PCT.
             ASPHALT
                                                        7.527
     ASPH. SPECIFIC BRAV.
                                                        1.023
     A66.
             SPECIFIC GRAV.
                                                        2.660
             AIR VOIDS ,PERCENT
VOL. CONCENTRATION -CALCULATED
                                                        6.200
     MIX.
     AGG.
                                                          .810
     COEF. OF CONTRACTION
                                   TEMP(F)
                                                   ALPH(10==5)
                                   -70.
                                                       1.000
                                     ٥.
                                                       1.200
                                                       1.400
                                    70.
                                   210.
                                                       1.800
     COEF. OF VARIATION OF ALPH
                                                        .100
     MAX. TEN.STRENGTH ,PSI
COEF. OF VARIATION OF MAX.STRENGTH
                                                      435.000
                                                         . 200
                            INPUT FATIGUE DATA
                                      N=A+(1.0/STRAIN)++B
             FATIBUE CURVE
     MIN.STIF.(PSI)
                             CONST.A
                                                   COMST.B
         .1000E+02
                              .1000E-01
                                                   .3000E+01
                              .8000E-12
         .5000E+07
                                                   .3950E+01
1 PAV.SEC.NO.
                           SIERRA ARNY BASE
                  - 1
```

Figure 11. Data used in temperature cracking program.

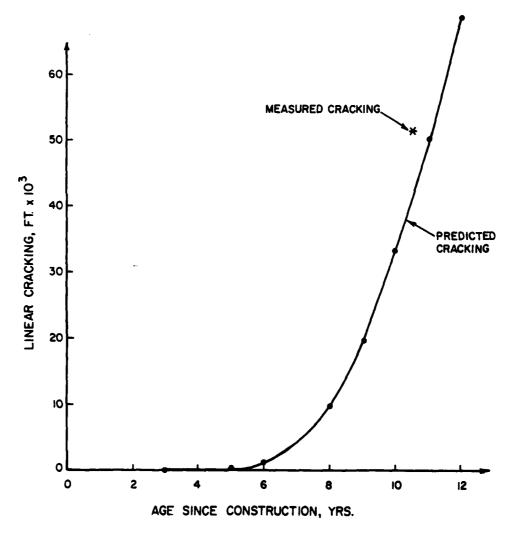


Figure 12. Predicted cracking using temperature cracking program.

	ility: Sierra Army Kunway Feature:		
l.	Overall Condition Rating - PCI=61 -> M4	2 Zone	= Koutine, Major, or Overall
Exc	ellent, Very Good, Good Fair, Poor, Very Poo	r, <u>Failed</u>	
2.	Variation of Condition Within Feature - PCI		
	a. Localized Random Variationb. Systematic Variation:	Wes,	No No
3.	Rate of Deterioration of Condition - PCI		
	a. Long-term period (since construction) b. Short-term period (1 year) Unknown	Low, Low,	Normal, High
4.	Distress Evaluation		
	a. Cause		
	Load Associated Distress Climate/Durability Associated Other () Associated Distress	percent	deduct values deduct values deduct values
	b. Moisture (Drainage) Effect on Distress	Minor,	Moderate, Major
5.	Load-Carrying Capacity Deficiency	No *	Yes
6.	Surface Roughness	Minor?	Moderate, Major
7.	Skid Resistance/Hydroplaning Unknoいん (runways only)	No hydr are exp	oplaning problems pected
	a. Mu-Meter		cional ial for hydroplaning igh probability
	b. Stopping Distance Ratio	Potent Potent	roplaning anticipated lal not well defined lal for hydroplaning ligh hydroplaning lal
	c. Transverse Slope	Poor,	Fair, Good, Excellent
8.	Previous Maintenance	low.	Normal, High
9.	Effect on Mission (Comments): * Povemon	1 13	Structurally
	adequate. However Aligates a	me Kin	assing 641

Figure 13. Airfield pavement condition evaluation summary.

M & R ZONE	PCI	RATING
CONTINE	100	EXCELLENT
ROUTINE	85	VERY GOOD
ROUTINE, MAJOR,	70	G000
OVERALL,	55	FAIR
MAJOR, OVERALL	40	POOR
OVERALL	25	VERY POOR
OVENALL	0	FAILED

Figure 14. Correlation of M&R zones with PCl and condition rating.

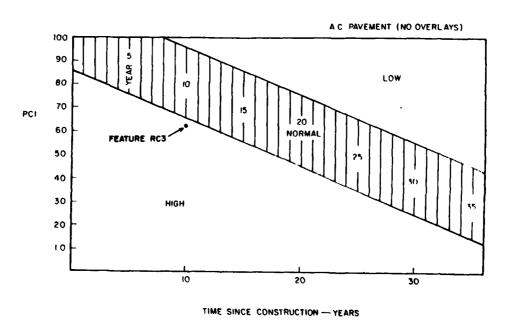


Figure 15. Rate of deterioration of AC pavements (no overlays).

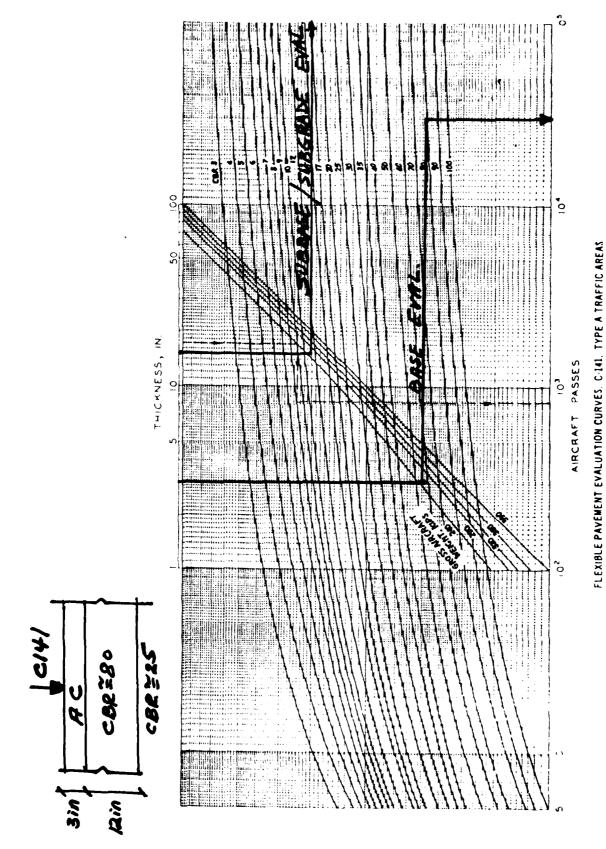


Figure 16. Evaluation of load carrying capacity.

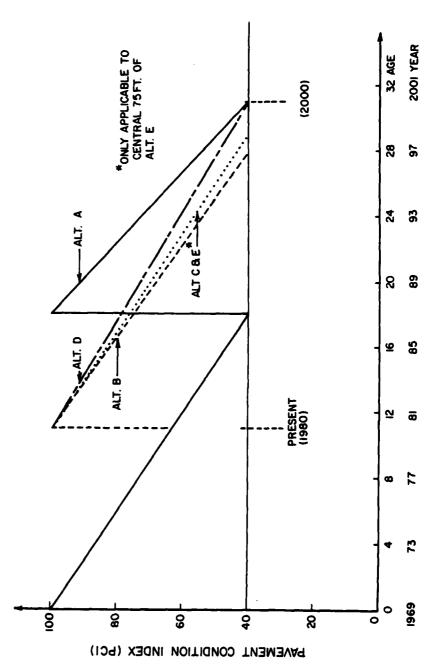
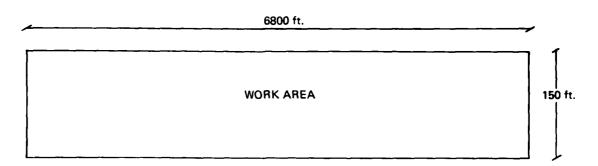


Figure 17. Expected PCI over time for each M&R alternative.

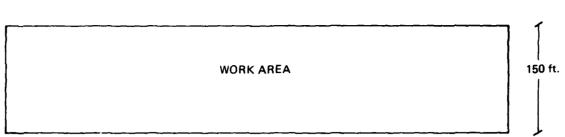


Work Summary: Continue to repair cracks as they appear or deteriorate.

Cracks less than 1 inch wide will be sealed. Cracks over 1 inch wide will be patched.

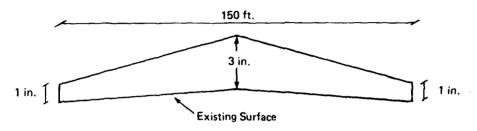
· ····································	-	
Initial Cost		
1. Seal cracks less than 1 inch wide	=	\$ 29,444
(See Est #1, Figure 19)		
2. Patch cracks over 1 inch wide	=	\$ 21,609
(See Est #2, Figure 19)		
3. Patch alligator and slippage cracking	=	\$ 5,361
(See Est #3, Figure 19)		
4. Apply rejuvenator		
$(\approx 0.1 \text{ gal/SY}, \approx \$0.4/\text{SY})$	Ξ	\$ 45,333
6800 x 150 x 1/9 x 0.4		
	Total Initial Cost	\$101,747
	Total Initial Cost	4101,747

Figure 18. Alternative A: continue crack seal to PCI=40 (1987), then overlay.



Work Summary: 1. Repair cracks (sealing & patching).

2. Overlay with AC as shown below.



Initial Cost

1. Seal all cracks less than 1 inch wide. Narrow cracks should be routed and cleaned. A space backer should be inserted before filling the cracks with a sealer. Assume half the medium severity cracks are less than 1 inch wide.

Total length of cracks to be sealed = 16588 + 1/2 (25712) = 29444LFCrack seal = $29444 \times 1.0 / LF$

2. Saw cut and patch cracks over 1 inch wide. The patch should be approximately 6 inches wide and

3 inches deep. Quantity = 1/2 (25712) + 8753 = 21609LF

Crack patch = $21609 \times 1.0 / LF$

\$ 21,609

3. Patch alligator and slippage cracking with 3-inch AC $(248SF + 5051SF + 62SF) \times $1.0/SF$ \$ 5,361

4. Tack coat — \$1.0/gal, Apply 0.1 gal/SY = @ \$0.1/SY

6800 x 150 x 1/9 x .1 \$ 11,333

5. New AC hot mix in place @ \$45/ton $(6800 \times 150 \times 2/12) \times 142 \times 1/2000 \times 45$

\$543,150

6. Apply rejuvenator, construction coat

 $(\simeq .075 \text{ gal}/\text{SY} \simeq \$0.3/\text{SY})$ 6800 x 150 x 1/9 x 0.3

\$ 34,000

\$ 29,444

Total Initial Cost

\$644,897

Figure 19. Alternative B: overlay 1980.

WORK AREA 150 ft.

Work Summary: 1. Cold mill surface and store on base for future use.

2. Place new 3 inch hot AC.

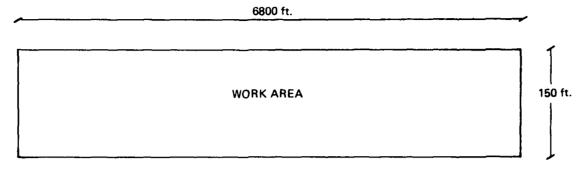
Initial Cost

1. Cold mill AC surface @ \$0.75/SY/in. \$ 255,000 6800 x 150 x 1/9 x 3 x 0.75 2. Hauling and stockpiling cold-milled material Assume 1-mile haul @ \$0.5/ton mile 6800 x 150 x 3/12 x 142 x 1/2000 x 0.5 9,053 3. Prime base course \$1.0/gal, Apply 0.2 gal/SY 6800 x 150 x 1/9 x 0.2 x 1.0 22,667 4. New AC hot mix @ \$45.0/ton 6800 x 150 x 3/12 x 142 x 1/2000 x 45 \$ 814,725 5. Apply rejuvenator, construction coat $(\simeq .075 \text{ gal/SY} \simeq \$0.3/\text{SY})$ $6800 \times 150 \times 1/9 \times .3$ 34,000

Total Initial Cost

\$1,135,445

Figure 20. Alternative C: replace surface — 3-inch deep.



Work Summary: 1. Cold mill 150 ft width.

- 2. Recycle cold milled material and reuse as stabilized base.
- 3. Place new AC hot mix as shown below.

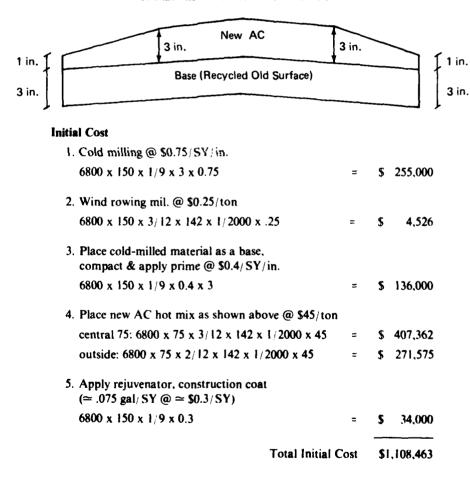
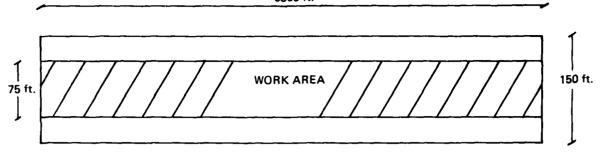


Figure 21. Alternative D: recycle surface and use as base, add new surface.



- Work Summary: 1. Cold mill central 75 ft -3 in. depth and store on base for future use.
 - 2. Place new 3 in. AC in central 75 ft.
 - 3. Maintain cracks in outside edges.

Initial Cost

1. Cold mill central 75 ft — 3 inch @ \$0.75/SY/in. 6800 x 75 x 1/9 x 3 x .75	z	\$127,500
2. Hauling and stockpiling cold-milled material assume 1 mile haul @ \$.05/ton mile		
6800 x 75 x 3/12 x 142 x 1/2000 x .5	=	\$ 4,526
3. Prime base course \$1.0/gal, Apply 0.2 gal/SY		
6800 x 75 x 1/9 x .2 x 1.0	=	\$ 11,333
4. New AC hot mix @ \$45.0/ton		
6800 x 75 x 3/12 x 142 x 1/2000 x 45	=	\$407,362
5. Repair cracks outside central 75 ft 18906 LF of crack x \$1.0/LF (See Table 4)	=	\$ 18.906
18700 ET OF CIACK X \$1.0/ ET (See Table 4)	-	J 10,700
6. Apply rejuvenator, central 75 ft (≈ .075 gal/SY @ ≈ \$0.3/SY)		
6800 x 75 x 1/9 x .3	=	\$ 17,000
7. Apply rejuvenator, outside 75 ft (≈ 0.1 gal/SY @ ≈ \$0.4/SY)		
6800 x 75 x 1/9 x .4	=	\$ 22,666
Total Initia	Cost	\$609,293

Figure 22. Alternative E: replace surface central 75 feet, crack seal outside.

REPORT BATE - 80/08/08.

COMPARISON OF MAR ALTERNATIVES SIERRA SECTION RU

ANALYSIS PERIOD	- 20 YEARS INTEREST RATE	
ALTERNATIVE	DESCRIPTION NET	PRESENT COST
3	OVERLAY-1980	945563.
E	REPLACE SURFACE CENTERAL 75 FT, CRK SEAL OUTSIDE	1007500.
٨	CONT CRK SEAL TO PCI=40(1987)THEN OVERLAY	1044139.
)	RECYCLE SURFACE AND USE AS DASE, ABD NEW SURFACE	1149794.
C	REPLACE SURFACE-3 INCH DEEP	1267111.

BETAILED COMPARISON OF MAR ALTERNATIVES

			ALT	A		ALT	1		ALT	C	•	ALT	D	٠	ALT	E	•
				PRES			PRES			PRES		1	PRES			PRES	
YEA	R		COST	COST		COST	COST		COST	COST		COST	COST	٠	COST	COST	
					٠							1					•
0	(FY80)		101747	101747		644897	644897	•	1135445	1135445		1108463	1108463	•	407273	609293	•
1	(FY81)		٥	Q	*	0	0		0	9		• 0	•		0	0	
2	(FY\$2)	*	41403	41402		10000	10000		0	0		• 0	•		17473	17473	•
3	(FY83)		0	0		•	0		•	0		• 0	0		•	•	•
4	(FY84)		43213	43212		20000	20000		0	0		•	•	•	21392	21392	
5	(FY65)		9	0	•	٥	٥		•	0	4	. 0	•	•	0	0	
6	(FY84)		83740	83960		30000	29999		0	0		• •	0		39002	37001	•
7	(FY87)		588483	588482		0	٥		9	0	•	• 0	0	•	0	•	
8	(FY88)		•	0	*	75333	75333		46333	46333		• 0	0	*	77819	77819	
9	(FY89)		10000	9979		0	0	•	0	. 0	•	•	0		0	0	
10	(FY90)		•	0		20000	20000		•	0		46333	44332	٠	27447	29448	•
11	(FY91)		20000	20000		0	0		0	0		• 0	0	٠	0	0	
12	(FY72)		0	0	•	30000	27777		5000	5000	4	. 0	•		44925	44725	•
13	(FY93)		30000	29999		0	0	•	•	0		. 0	0		0	0	
14	(FY94)	٠	0	0	*	30000	30000	•	10000	10000	•	5000	5000	•	38103	38103	٠
15	(FY95)		75333	75333		0	0	•	0	0	•	• •	0		0	0	
14	(FY94)		•	0		75333	75333		60333	40333		. 0	0		82544	82544	•
17	(FY97)		20000	20000		•	•	•	0	•		• 0	0	٠	0	0	•
18	(FY98)		•	0		30000	30000		10000	9999		10000	7777	•	47478	47477	•
19	(FY99)		30000	30000		•	0	•	0	0		• 0	0	٠	0	0	•
20	(FYOO)		0	0		0	•		0	0	•	• 0	0	٠	0	0	٠
								•)		٠			•
TO	TAL		1044137	1044138		745543	745543	•	1247111	1267111	•	1149796	1169796	•	1007500	1007500	•
		•						•			•	1		•			•
SA	LVAGE	•	0	0		0	•	•	0	0	•	•	•		•	0	•
		•			•							1		٠			•
PRES	VORTH			1044138			745543	•		1267111	•)	1149794	•		1007500	•

Figure 23. Life cycle costing of M&R alternatives.

Table 1
Summary of Linear Cracking for Entire Runway

Feature ID	Low Sev.		Med. Sev.		High Sev.		Total Crk.	
	Quant. LF	Dens. %	Quant. LF	Dens. %	Quant. LF	Dens. %	Quant. LF	Dens %
RN1	215	3.82	44	.78			259	4.6
RN2	1348	.54	6506	2.6	2473	.99	10327	4.13
RSI	37	.65					37	0.65
RS2	3274	1.31	2666	1.06	2343	0.93	8283	3.33
RCI	328	2.91	329	2.92			657	5.83
RC2	1571	2.46	1030	1.61	800	1.25	3401	5.32
RC3	7711	2.05	12762	3.4	3074	.81	23547	6.26
RC4	2104	3.50	2375	3.95	63	0.1	4542	7.55
Total				•				
Crk. &	16588	1.63	25712	2.52	8753	0.86	51053	5.01
Overall								
Dens.								

Low Severity Cracking, C_i of Total Cracking = (16588-51053) x 100 = 32.5 C_i Med. Severity Cracking, C_i of Total Cracking = (25712-51053) x 100 = 50.4 C_i High Severity Cracking, C_i of Total Cracking = (8753-51053) x 100 = 17.1 C_i

Table 2
Summary of Distresses Other Than Linear Cracking

Feature ID	Distress	Low Sev.		Med. Sev.		High Sev.		Total Crk.	
		Quant.	Dens.	Quant.	Dens.	Quant.	Dens.	Quant.	Dens.
RC2	Slippage* Cracking							248SF	0.38
RC3	Alligator Cracking	348	.09	3681	0.98	1022	.27	5051SF	1.34
RC4	Alligator Cracking			62	0.1			6281	.1

^{*}Slippage cracking has no severity levels.

Table 3
Predicted Cracks and Repair Costs for Entire Runway

Age Years	Year	Total Cracks, LF	Cracks to be Sealed, LF	Cracks to be Resealed, LF	Cost of Crack Repair (@ \$1.0/LF)
tt	80	51053	51053	0	51053
13	82	92456	41403	0	41403
15	84	135669	43213	0	43213
17	86	168576	32907	51053	83960
19	88	186863	18287	41403	59690
21	90	194291	7428	43213	50641
23	92	196498	2207 ·	83960	86167
25	94	196971	473	59690	60163
27	96	197030	59	50641	50700
29	98	197050	20	86167	86187
31	00	197050	0	60163	60163

\$673,340

Table 4
Summary of Linear Cracking Outside the Central 75 Ft

Feature	Low	Low Sev. Med. Sev.	High	High Sev.		Total Crk.		
•	Quant. LF	Dens. %	Quant. LF	Dens. %	Quant. LF	Dens. %	Quant. LF	Dens.
RNI	215	3.82	44	.78			259	4.6
RN2	1348	.54	6506	2.6	2473	.99	10327	4.13
RS1	37	.65					37	.65
RS2	3274	1.31	2666	1.06	2343	.93	8283	3.33
Total	····	···						
Crk. &	4874	0.96	9216	1.81	4816	0.94	18906	3.71
Overall								
Dens.								

Low Severity Cracking, % of Total Cracking = (4874/18906) = 25.8

Med. Severity Cracking, % of Total Cracking = (9216/18906) = 48.7

High Severity Cracking, % of Total Cracking = (4816/18906) = 25.5

Table 5
Predicted Cracks and Repair Cost for Outside the Central 75 Ft

Age Years	Year	Total Cracks, LF	Cracks to be Sealed, LF	Cracks to be Resealed, LF	Cost of Crack Repair (@ \$1.0/LF)
11	80	18906	18906	0	18906
13	82	36379	17473	0	17473
15	84	57771	21392	0	21392
17	86	77867	20096	18906	39002
19	88	92381	14514	17473	31987
21	90	100438	8057	21392	29449
23	92	103861	3423	39002	42425
25	94	104977	1116	31987	33103
27	96	105262	285	29449	29734
29	98	105315	53	42425	42478
31	00	105325	10	33103	33113

\$339,062

Table 6
Comparison of S/Unit Performance for Each M&R Alternative

Net Present Cost A 1,044,138		Area Between PCI & Min. PCI			S/Unit Performance	
		(63 - 40) x 7/2 +				
		(100 - 40) x 13/2	=	470.5	2219	
В	965,563	$(100 - 40) \times 17/2 + 0$	=	510	1893	
C.	1,267,111	$(100 - 40) \times 18/2 + 0$	•	540	2346	
D	1,169,796	(100 40) x 20/2		600	1950	
E*	1,007,500	(100 - 40) x 18 2	-	540	1866	

^{*}Only applicable to the central 75 ft.

0 FEATURE IDENTIFICATION = RN1 SIERRA AFB CA DATE SURVEYED 11/14/79. FLEXIBLE PAVENENT. FEATURE SIZE = 00005625 SF TOTAL NO OF SAMPLE UNIT = 1 ALLOWABLE ERROR WITH 95% CONFIDENCE = 5 SAMPLE UNIT ID = AREA OF SAMPLE, SF = 5625 DENSITY Z DEDUCT VALUE SEVERITY DISTRESS-TYPE QUANTITY 3.82 11.5 08 LON 215 08 HEBIUM 0.78 10.0 PCI = 87 NO. OF RANDOM SAMPLE = NO. OF ADDITIONAL SAMPLE = PCI OF FEATURE -RN1 SIERRA AFB CA = 89 RATING = EXCELLENT RECOMMEND EVERY SAMPLE UNITS TO BE SURVEYED. ESTINATED DISTRESS FOR FEATURE = RN1 SIERRA AFD CA DISTRESS-TYPE SEVERITY YTITHAUD DENSITY I DEDUCT VALUE LOW 08 215 3.82 11.5 08 MEBIUN 44 0.78 10.0

FEATURE IDENTIFICATION = RM2 SIERRA AFB CA

DATE SURVEYED 11/14/79.

FLEXIBLE PAVENENT.

FEATURE SIZE =

00249375 8F

TOTAL NO OF SAMPLE UNIT =

45

ALLOWABLE ERROR WITH 95% CONFIDENCE = 5

SAMPLE UNIT ID = AREA OF SAMPLE, SF = 5625

BENSITY & DEBUCT VALUE YTITHAUS DISTRESS-TYPE SEVERITY 19.8 MEDIUM 170 3.02 08

PCI = 80

SAMPLE UNIT IB = AREA OF SAMPLE, SF = 5625

QUANTITY DENSITY Z DEDUCT VALUE BISTRESS-TYPE SEVERITY MEDIUM 150 2.66 18.5 98

PCI . 82

16 SAMPLE UNIT IB = AREA OF SAMPLE, SF = 5625

DENSITY % DEBUCT VALUE SEVERITY YTITKAUD DISTRESS-TYPE 2.18 8.3 02 LOU 123 MEDIUM 96 1.70 15.0 98

PCI = 85

SAMPLE UNIT ID = AREA OF SAMPLE, SF = 5625

YTITHAUD DENSITY Z DEDUCT VALUE DISTRESS-TYPE SEVERITY 49 0.87 4.9 80 LOU 1.95 16.0 110 08 MEDIUM

PCI = 79

SAMPLE UNIT IB = AREA OF SAMPLE, SF = 5625

DENSITY Z DISTRESS-TYPE SEVERITY PTITHAUD DEDUCT VALUE 0.44 98 3.4 LOW 25 80 2.00 16.3 MEDIUM 113

PCI = 80

SAMPLE UNIT ID = 32 *ADDITIONAL*
AREA OF SAMPLE,SF = 5625

DISTRESS-TYPE	SEVERITY	QUANTITY	DENSITY Z	DEDUCT VALUE
98	LOW	110	1.95	7.8
98	HEDIUN	37	0.65	9.1
08	HIGH	47	0.83	18.3

PCI = 82

SAMPLE UNIT ID = 34 AREA OF SAMPLE, SF = 5625

DISTRESS-TYPE	SEVERITY	QUANTITY	DENSITY Z	DEDUCT VALUE
08	LOW	3	0.05	1.2
08	HEDIUH	138	2.45	17.8
08	HIGH	392	6.96	46.3

PCI = 54

SAMPLE UNIT ID = 40 AREA OF SAMPLE, SF = 5625

DISTRESS-TYPE	SEVERITY	QUANTITY	DENSITY Z	DEDUCT VALUE
08	HEDIUH	268	4.76	24.3

PCI = 76

NO. OF RANDOM SAMPLE = 7

NO. OF ADDITIONAL SAMPLE = 1

PCI OF FEATURE -RN2 SIERRA AFB CA = 77 RATING = V. 800D

RECOMMENDED MINIMUM OF 17 RANDOM SAMPLE UNITS TO BE SURVEYED.

STANDARD DEVIATION OF PCI BETWEEN RANDOM UNITS SURVEYED= 10.3

ESTINATED DISTRESS FOR FEATURE = RM2 SIERRA AFB CA

DISTRESS-TYPE	SEVERITY	PTITHAUD	DENSITY I	DEDUCT VALUE
08	LOU	1348	0.54	3.9
08	HEDIUN	6506	2.60	18.3
08	HIGH	2473	0.99	19.9

٥

0

0

FEATURE IDENTIFICATION = RS1 SIERRA AFB CA

BATE SURVEYED 11/14/79. FLEXIBLE PAVEMENT.

FEATURE SIZE = 00005625 SF

TOTAL NO OF SAMPLE UNIT =

ALLOWABLE ERROR WITH 95% CONFIDENCE = 5

SAMPLE UNIT ID = 1 AREA OF SAMPLE, SF = 5625

DISTRESS-TYPE SEVERITY QUANTITY DENSITY % DEDUCT VALUE 08 LOW 37 0.65 4.2

PCI = 96

NO. OF RANDOM SAMPLE =

NO. OF ADDITIONAL SAMPLE = 0

PCI OF FEATURE -RS1 SIERRA AFB CA = 96 RATING = EXCELLENT

RECOMMEND EVERY SAMPLE UNITS TO BE SURVEYED.

ESTINATED DISTRESS FOR FEATURE = RS1 SIERRA AFB CA

DISTRESS-TYPE SEVERITY QUANTITY DENSITY I DEDUCT VALUE 08 LOW 37 0.65 4.2

38

FEATURE IDENTIFICATION = RB2 SIERRA AFB CA

DATE SURVEYED 11/14/79. FLEXIBLE PAVENENT.

FEATURE SIZE = 00249375 SF

TOTAL NO OF SAMPLE UNIT = 45

ALLOWABLE ERROR WITH 95% CONFIDENCE = 5

SAMPLE UNIT ID = 4
AREA OF SAMPLE, SF = 5625

DISTRESS-TYPE	SEVERITY Lou	QUANTITY 2	DENSITY Z	DEDUCT VALUE
08	MEBIUM	•	0.10	4.0
08	HISH	126	2.24	29.0

PCI = 66

SAMPLE UNIT ID = 10 AREA OF SAMPLE, SF = 5625

DISTRESS-TYPE	SEVERITY	YTITHAUD	DENSITY Z	BEDUCT VALUE
80	LOU	26	0.46	3.7
08	HIBH	74	1.35	22.9

PCI = 73

SAMPLE UNIT ID = 16 AREA OF SAMPLE, SF = 5625

BISTRESS-TYPE	SEVERITY	YTITHAUD	BENSITY I	BEBUCT VALUE
08	LOU	<i>7</i> 3	1.29	6.0
98	MEDIUM	106	1.88	15.7

PCI = 84

SAMPLE UNIT ID = 22 AREA OF SAMPLE.SF = 3625

DISTRESS-TYPE	SEVERITY	PTITHAUD	DENSITY X	DEDUCT VALUE
08	LOW	194	3.48	10.9
08	MEDIUM	88	1.56	14.4
08	HIGH	38	0.47	16.4

PCI = 78

SAMPLE UNIT IB = AREA OF SAMPLE,SF				
DISTRESS-TYPE	SEVERITY	QUANTITY	DENSITY I	DEDUCT VALUE
08	LOU	46	0.81	4.7
08	MEDIUM	37	0.65	7.1
08	HIGH	37	0.65	16.3
			PCI = 83	
SAMPLE UNIT ID =	34			
AREA OF SAMPLE, SF				
DISTRESS-TYPE	SEVERITY	QUANTITY	DENSITY I	DEBUCT VALUE
08	LOU	3	0.05	1.2
08	MEDIUM	147	2.61	18.3
08	HIGH	55	0.97	19.7
			PCI = 76	
SAMPLE UNIT ID = AREA OF SAMPLE,SF				
BISTRESS-TYPE	SEVERITY	QUANTITY	DENSITY %	DEDUCT VALUE
08	LOW	171	3.04	10.0
08	HEBIUM	37	0.65	9.t
98	High	38	0.67	16.6
			PCI - 82	
NO. OF RANBON SAMPLE =	1	7		
NO. OF ABDITIONAL SAMP	LE =	•		
PCI OF FEATURE -RS2 9	IERRA AFB C	A = 77	RATING =	V. 6003
RECONNENDED MINIMUM OF	9 RANDO	N SAMPLE UNITS	TO BE SURVEYE	D.
ABAMBABB ABMBABBB 35				
STANDARD DEVIATION OF	PCI BETUEEN	RANDOR UNITS S	SURVEYEDZ 4.	4
ESTINATED DISTRESS F	OR FEATURE	- RS2 SIERRA	AFB CA	

·

SEVERITY LOW NEDIUM

HIGH

DISTRESS-TYPE 08 08 08

DENSITY X DEDUCT VALUE 6.1

11.8

19.3

1.06

FEATURE IDENTIFICATION = RC1 SIERRA AFB CA

DATH SURVEYED 11/14/79. FLEXIBLE PAVENENT.

FEATURE SIZE X 00011250 SF

TOTAL NO OF SAMPLE UNIT 2

ALLOWABLE ERROR WITH 75% CONFIDENCE = 5

SAMPLE UNIT IB = 1 AREA OF SAMPLE, SF = 5625

DISTRESS-TYPE	SEVERITY	QUANTITY	DENSITY Z	DEDUCT VALUE
08	FOR	183	3.25	10.4
08	MEDIUM	289	5.13	25.2

PCI = 75

SAMPLE UNIT ID = 2 AREA OF SAMPLE, SF = 5625

DISTRESS-TYPE	SEVERITY	GUANTITY	BENSITY I	BEDUCT VALUE
08	FOA	145	2.57	9.1
08	HEDIUN	40	0.71	9.5

PCI = 91

NO. OF RANDON SAMPLE =

NO. OF ADBITIONAL SAMPLE = 0

PCI OF FEATURE -RC1 SIERRA AFD CA = 83 RATING = V. 600D

RECOMMEND EVERY SAMPLE UNITS TO BE SURVEYED.

ESTINATED DISTRESS FOR FEATURE = RC1 SIERRA AFB CA

DISTRESS-TYPE	SEVERITY	QUANTITY	DENSITY Z	DEDUCT VALUE
08	LOW	328	2.91	9.8
08	MEBIUM	329	2.92	19.5

Λ

FEATURE IDENTIFICATION = RC2 SIERRA AFD CA

BATE SURVEYED 11/14/79.

FLEXIBLE PAVEMENT.

FEATURE SIZE %

00063750 BF

TOTAL NO OF SAMPLE UNIT 2 11

ALLOWABLE ERROR WITH 951 CONFIDENCE = 5

SAMPLE UNIT ID = 1 AREA OF SAMPLE, SF = 5625

BISTRESS-TYPE	SEVERITY	PTITHAUD	DENSITY Z	DEDUCT VALUE
08	LOW	363	6.45	16.9
08	NEDIUN	97	1.72	15.1

PCI = 81 .

SAMPLE UNIT ID = 3 AREA OF SAMPLE, SF = 5625

BISTRESS-TYPE	SEVERITY	YTITHAUD	DENSITY 2	DEDUCT VALUE
08	LOV	95	1.68	7.0
08	HEDIUM	64	1.13	12.2
08	HIGH	88	1.54	24.6

PCI = 75

SAMPLE UNIT IB = AREA OF SAMPLE, SF = 5625

DISTRESS-TYPE	SEVERITY	YTITHAUD	DENSITY Z	DEBUCT VALUE
98	LOW	145	2.57	9.1
08	MEDIUM	102	1.81	15.5
0.0	MICH	30	0.53	14.R

PCI = 80

SAMPLE UNIT IB = 7 AREA OF SAMPLE, SF = 5625

DISTRESS-TYPE	SEVERITY	QUANTITY	DENSITY Z	DEDUCT VALUE
08	LOW	44	0.78	4.6
08	HEDIUM	93	1.65	14.8
08	HIGH	94	1.67	25.4

PCI = 71

SAMPLE UNIT ID = 9 *ADDITIONAL* AREA OF SAMPLE, SF = 5625

DISTRESS-TYPE	SEVERITY	YTITHAUD	DENSITY Z	DEBUCT VALUE
08	LOU	23	0.40	3.5
08	NEDIUN	114	2.02	14.3
08	HIGH	142	2.52	30.5
15		132	2.34	21.9

PCI = 56

SAMPLE UNIT ID = 11 AREA OF SAMPLE, SF = 7500

BISTRESS-TYPE	SEVERITY	PUANTITY	DENSITY Z	DEDUCT VALUE
08	LOW	152	2.02	8.0
08	MEDIUM	117	1.56	14.4
08	HIGH	128	1.70	25.6
15		60	0.80	10.5

PCI = 72

NO. OF RANDON SAMPLE = 5

NO. OF ADDITIONAL SAMPLE = 1

PCI OF FEATURE -RC2 SIERRA AFB CA = 74 RATING = V. GOOD

RECONNENDED MINIMUM OF 005 RANDOM SAMPLE UNITS TO BE BURVEYED.

STANDARD DEVIATION OF PCI DETUEEN RANDON UNITS SURVEYEDZ 4.5

ESTINATED DISTRESS FOR FEATURE = RC2 SIERRA AFB CA

DISTRESS-TYPE	SEVERITY	QUANTITY	DENSITY 2	DEDUCT VALUE
08	LOW	1571	2.46	8.9
08	NEBIUN	1030	1.61	14.6
08	HIGH	800	1.25	22.1
15		248	0.38	7.3

0

FEATURE IDENTIFICATION = RC3 SIERRA AFB CA

BATE SURVEYED 11/14/79.

FLEXIBLE PAVENENT.

FEATURE SIZE Z

00375000 SF

TOTAL NO OF SAMPLE UNIT X 67

ALLOWABLE ERROR WITH 95% CONFIDENCE = 5

SAMPLE UNIT ID = AREA OF SAMPLE, SF = 5625

DISTRESS-TYPE	SEVERITY	YTITHAUD	DENSITY Z	DEBUCT VALUE
08	LOU	152	2.70	9.4
08	HEDIUM	78	1.38	13.5
08	HIGH	77	1.34	23.0

PCI = 75

SAMPLE UNIT IB = 13 AREA OF SAMPLE, 8F = 5625

DISTRESS-TYPE	SEVERITY	QUANTITY	DENSITY X	DEBUCT VALUE
08	FOR	140	2.48	8.9
08	HEDIUM	74	1.31	13.1
98	HBH	95	1.68	25.4

PCI = 74

SAMPLE UNIT ID = AREA OF SAMPLE, SF = 5625

DISTREBS-TYPE	SEVERITY	YTITHAUS	DENSITY X	DEDUCT VALUE
08	LOU	182	3.23	10.4
80	NEDIUN	206	3.66	21.5
08	HIGH	76	1.35	22.9
01	LOU	18	0.32	11.1
01	NEBIUM	225	4.00	44.0
01	HISH	24	0.42	28.3

PCI = 33

SAMPLE UNIT ID =	27			
AREA OF SAMPLE, SF	= 5625			
DISTRESS-TYPE	SEVERITY	QUANTITY	DENSITY Z	DEBUCT VALUE
08	LOW	55	0.97	5.3
08	NEDIUN	183	3.25	20.4
08	High	76	1.35	22.7
01	MEDIUM	17	0.30	10.5
01	HIGH	3	0.05	8.0
			PCI = 61	
SAMPLE UNIT ID =	34			
AREA OF SAMPLE, SF	= 5625			
DISTRESS-TYPE	SEVERITY	YTITHAUD	DENSITY Z	DEDUCT VALUE
08	LOW	61	1.44	6.3
08	HEDIUM	271	4.81	24.5
08	HIGH	69	1.22	21.8
01	LOW	10	0.17	8.1
01	MEDIUM	5	0.98	8.0
01	HIGH	45	0.80	34.3
			PCI = 47	
SAMPLE UNIT ID =	41			
AREA OF SAMPLE, SF				
DISTRESS-TYPE	SEVERITY	PTITHAUD	DENSITY Z	DEDUCT VALUE
08	LOW	150	2.66	9.3
08	WEDIUM	182	3.23	20.3
08	HIGH	10	0.17	7.0
			PCI = 80	
SAMPLE UNIT ID = AREA OF SAMPLE,SF				
DISTRESS-TYPE		A	DEMOTTY W	DEDUCT VALUE
	SEVERITY	QUANTITY	DENSITY Z	
08	LOW	57	1.01	5.5
08 08 01				

SAMPLE UNIT ID = 55 AREA OF SAMPLE.SF = 5625

DISTRESS-TYPE	SEVERITY	PITTHAUS	DENSITY Z	DEDUCT VALUE
08	LOW	182	3.23	10.4
08	MEDIUN	262	4.45	24.0
01	LOU	19	0.33	11.3
0 1	NEDIUN	10	0.17	13.8
01	HIGH	44	1.17	30.5

PCI = 49

SAMPLE UNIT ID = 62 AREA OF SAMPLE, BF = 5625

DISTRESS-TYPE	SEVERITY	QUANTITY	DENSITY Z	DEDUCT VALUE
08	LOW	42	0.74	4.4
08	NEBIUM	293	5.20	25.4
08	HIGH	12	0.21	10.0

PCI = 75

NO. OF RANDOM SAMPLE =

NO. OF ADDITIONAL SAMPLE = 0

PCI OF FEATURE -RC3 SIERRA AFB CA = 61 RATING = 800D

RECOMMENDED MINIMUM OF 31 RANDOM SAMPLE UNITS TO DE SURVEYED.

STANDARD DEVIATION OF PCI DETWEEN RANDON UNITS SURVEYEDX 16.1

ESTINATED DISTRESS FOR FEATURE - RC3 SIERRA AFD CA

DISTRESS-TYPE	SEVERITY	PUANTITY	DENSITY I	DEDUCT VALUE
01	LOU	348	0.09	4.3
Q1	HEDIUH	3481	♦.78	29.2
01	HIGH	1022	0.27	24.2
08	FOR	7711	2.05	8.1
08	HEBIUH	12742	3.40	20.8
08	HIGH	3074	0.81	18.1

46

FEATURE IDENTIFICATION = RC4 SIERRA AFB CA

BATE SURVEYED 11/13/79.

FLEXIBLE PAVENENT.

FEATURE SIZE Z

00040000 SF

TOTAL NO OF SAMPLE UNIT % 11

ALLOWABLE ERROR WITH 95% CONFIDENCE = 5

SAMPLE UNIT ID = AREA OF SAMPLE, SF = 5625

BISTRESS-TYPE	SEVERITY	QUANTITY	DENSITY X	DEBUCT VALUE
80	LOW	120	2.13	8.2
08	NEDIUN	217	3.89	22.1
01	MEDIUM	12	0.21	15.4

PCI = 75

SAMPLE UNIT ID = AREA OF SAMPLE, SF = 5625

DISTRESS-TYPE	SEVERITY	YTITHAUD	DENSITY Z	DEDUCT VALUE
08	LOW	233	4.14	12.1
08	NEDIUM	291	5.17	25.3
01	MEDIUM	35	0.62	24.8

PC1 = 43

SAMPLE UNIT ID = AREA OF SAMPLE, SF = 5625

BISTRESS-TYPE	SEVERITY	QUANTITY	DENSITY Z	DEDUCT VALUE
08	LOW	221	3.92	11.7
08	NEDIUN	157	2.79	19.0

PCI = 81

RAMPLE UNIT IB = 7 AREA OF SAMPLE, SF = 5625

DISTRESS-TYPE	SEVERITY	QUANTITY	DENSITY I	DEBUCT VALUE
08	LOW	45	1.15	5.8
08	NEDIUN	190	3.37	20.7
08	HIGH	63	1.12	21.0
01	MEDIUM	15	0.26	17.3

PC1 = 68

SAMPLE UNIT ID = 9 AREA OF SAMPLE, SF = 5625

DISTRESS-TYPE	BEVERITY	PTITHAUG	DENSITY Z	DEDUCT VALUE
08	LOU	217	3.89	11.6
98	Hebiuh	280	4.97	24.9

PCI = 75

NO. OF RANDOM SAMPLE =

NO. OF ADDITIONAL SAMPLE = 0

PCI OF FEATURE -RC4 SIERRA AFB CA = 72 RATING = V. SOOB

RECOMMENDED MINIMUM OF 7 RANDOM SAMPLE UNITS TO BE SURVEYED.

STANDARD DEVIATION OF PCI DETWEEN RANDON UNITS SURVEYEDZ 4.9

ESTINATED DISTRESS FOR FEATURE = RC4 SIERRA AFB CA

BISTRESS-TYPE	SEVERITY	TITHAUD	DENSITY Z	DEDUCT VALUE
01	MEDIUM	132	0.22	15.8
08	LOU	1830	3.05	10.0
08	MEDIUM	2425	4.04	22.5
08	HISH	134	0.22	10.2

1	FEATURE	PCI	RATING
	RC3 SIERRA AFB CA	61	600D
	RC4 SIERRA AFB CA	72	V. 800D
	RC2 SIERRA AFB CA	74	V. 600D
	RM2 SIERRA AFD CA	77	V. 600D
	R82 SIERRA AFD CA	77	V. 800B
	RC1 SIERRA AFB CA	83	V. GOOD
	RN1 SIERRA AFB CA	89	EXCELLENT
E01 (RS1 SIERRA AFB CA ENCOUNTERED.	76	EXCELLENT

APPENDIX B: RESULTS OF AC LABORATORY TESTING

Table B1 Bituminous Mix Analysis

PROJECT Sierra Army Depot		JOB NO		DATE 20 May 1980		
SOURCE		SAMPLED BY		DATE REC'D		
ESCRIPTION OF M	ATERIALS A	sphalt Cement	Slabs			
			T T	=======================================		
ABORATORY NO.	FPL 5864		 		+	
TELD NO.			1		 	
THER IDENTIFICA	TION		Top La	yer	Bottom	Layer
AVENENT CRITER			 		 	
) PSI TIRE PRES		JOB MIX	Laboratory	Field	Laboratory	Field
IZE OF	SPECIFIED LIMITS	FORMULA (APPROVED)	Samples*	Samples	Samples*	Sample
1 INCH			1		1	
3/4 INCH		<u> </u>	100.0		100.0	
1/2 INCH			94.3		93.7	
3/8 INCH			85.2		82.6	
NO. 4			71.6		63.8	
NO. 8			56,6		48.2	
NO. 16			41.0		35.2	
NO. 30			30.1		26.2	
NO. 50			21.6		19.0	
NO. 100			14.3		12.6	
NO. 200			8.6		7.6	
PERCENT			7.0	7.0	5.5	5.5
RADE						
STABILITY MARSHALL) LBS			5466		5153	
FLOW			16		17	
PERCENT VOIDS			4.9	6.2	8.9	10.4
PERCENT VOIDS			75.8	70.6	57-3	52.8
DENSITY - L85/CU	FT	11	142.4	140.2	140-6	138.0
THEO DENSITY - L		11	149.7		154.3	T
	63/CU F1	 				
STRIPPING, 3		 	 			
SWELL, "			2.66	 	2.69	
AGG - SP GR		 		 		
AGG - % WATER AB	SORPTION	<u></u>	1.6	<u> </u>	2.0	<u> </u>
REMARKS:		Test o	on Recovered .	Asphalt		
Penetration			15		13	
Softening Pt.	°c		71.2°C		75.0°C	
Viscosity	Poises	140°F	113,038		268,884	
	CST	225 ⁰ F	20,321		27,535	
	CST	225°F 275°F	2,230		3,451	
TESTED BY:		Guratom com	paction at 20	MO nai. on	e degree, and	l
CHECKED BY:			ns which is e			•
			paction effor			

Table B2
Indirect Tensile Results

Sample No.	Maximum Load (lb)	Vertical Deformation (in.)	Temp (°F)	Load Rate (in./min)
1	1950	0.080	75	2.0
l=1 top l=1 bottom	2400	0.095	75	2.0
11 tottom 12 top	2550	0.055	50	2.0
1-2 top 1-2 bottom	3000	0.048	50	2.0
	2800	0.045	20	2.0
I=3 top I=3 bottom	2200	0.065	20	2.0
	1025	0.095	75	0.05
2 - 1-top 2 - 1-bottom	900	0.120	75	0.05
= :	2470	0.080	50	0.05
2-2 top 2-2 bottom	1990	0.065	50	0.05
	3650	0.055	20	0.05
2–3 top 2–3 bottom	3450	0.042	20	0.05
Additional top	4020	0.042	-20°1	0.05
Additional top Additional bottom	4025	0.040	-20°F	0.05

Table B3 Sample Size

Sample No.	Average Height (in.)	Average Diameter (in.)
op I-l	1.064	3.951
1-2	1.066	3.957
1-3	1.690	3.960
op 2-1	1.485	3.960
2-2	1.587	3.966
2-3	1.587	3,955
Bottom 1-1	1.462	3,959
1-2	1.460	3,962
1-3	1.491	3,959
Bottom 2-1	1.601	3,962
2-2	1.561	3,949
2-3	1.577	3,957
T Additional	1.665	3,938
Top-Additional Bottom-Additional	1.723	3.917

Table B4 Computation of Indirect Tensile Strength

3.959

2200

237.27

Rate 2 in./min.	S = -	2 <u>P</u> g d			
Temp	Sample ID	<u> </u>	d	P, lbs	S,Psi
75°F	Top 1-1	1.064	3.951	1950	295.30
50°F	Top 1-2	1.066	3.957	2550	384.86
20°F	Top 1-3	1.69	3.96	2800	266.352
75°F	Bottom 1-1	1.462	3.959	2400	263.93
50°F	Bottom 1-2	1.460	3.962	3000	330.17

1.491

Rate .05 in./min.

50°F 20°F

Bottom 1-3

Temp	Sample 1D	Q	d	P, lbs	<u> </u>
75°F	Top 2-1	1.485	3.96	1025	110.96
50°F	Top 2-2	1.587	3.966	2470	249.83
20°F	Top 2-3	1.587	3.955	3650	370.2
-20°F	Top-additional	1.665	3.938	4020	390.32
75°F	Bottom 2-1	1.601	3.962	900	90.33
50°F	Bottom 2-2	1.561	3,949	1990	205.52
20°F	Bottom 2-3	1.577	3,957	3450	351,97
-20°F	Bottom-additional	1.723	3.917	4025	379.67

APPENDIX C: PROGRAM OUTPUT OF PCI PREDICTION FOR EACH M&R ALTERNATIVE

Alternative A. Overlay in 1987 — 2 inch average thickness.

SIERRA

C141 AIRCRAFT ID

O.O AGE BETWEEN ORIGINAL CONSTRUCTION AND LAST OVERLAY

3.0 TOTAL AC THICKNESS IN INCHES INCLUDING OVERLAYS

15.0 TOTAL PAVENENT THICKNESS ABOVE SUBBRADE

80.0 CBR OF BASE

25.0 CBR OF SUBGRADE

18.0 YEARS TO OVERLAY FROM LAST CONST/OVERLAY

2.0 THICKNESS OF OVERLAY

AGE SINCE OVERLAY	PCI
0.0	100.0
5.0	77.2
10.0	54.4
13.0	40.8

Alternative B. Overlay 1980 — 2-inch AC average thickness.

SIERRA

C141 AIRCRAFT ID

0.0 AGE BETWEEN ORIGINAL CONSTRUCTION AND LAST OVERLAY

3.0 TOTAL AC THICKNESS IN INCHES INCLUDING OVERLAYS

15.0 TOTAL PAVENENT THICKNESS ABOVE SUBGRADE

80.0 CBR OF BASE

25.0 CBR OF SUBBRADE

11.0 YEARS TO OVERLAY FROM LAST CONST/OVERLAY

2.0 THICKNESS OF OVERLAY

AGE SINCE OVERLAY	PCI
0.0	100.0
5.0	82.2
10.0	64.5
15.0	46.7
20.0	28.9

Alternatives C & E. Replace entire surface.

SIERRA

C141 AIRCRAFT IB

0.0 AGE BETWEEN ORIGINAL CONSTRUCTION AND LAST OVERLAY

3.0 TOTAL AC THICKNESS IN INCHES INCLUDING OVERLIYS

15.0 TOTAL PAVENENT THICKNESS ABOVE SUBSRADE

80.0 CBR OF BASE

25.0 CBR OF SUBGRADE

AGE SINCE LAST CONST/OVERLAY	PCI
0.0	100.0
5.0	83.3
11.0	63.3
20.0	33.4
25.0	16.7
31.0	0.0

Alternative D. Reuse surface as base and add new 3 inch AC.

SIERRA

C141 AIRCRAFT IB

Q.O AGE BETWEEN ORIGINAL CONSTRUCTION AND LAST OVERLAY

3.0 TOTAL AC THICKNESS IN INCHES INCLUDING OVERLAYS

18.0 TOTAL PAVENENT THICKNESS ABOVE SUBGRADE

100.0 CBR OF BASE

25.0 CBR OF SUBGRADE

AGE SINCE LAST CONST/OVERLAY	PCI
0.0	100.0
5.0	85.0
10.0	70.0
15.0	54.9
20.0	10.0

APPENDIX D: PROGRAM OUTPUTS FOR PREDICTING FUTURE CRACKING

DISTRESS INPUT DATA

DISTRESS	TYPE	=	8.			
ABE		11.00	YEARS			
L		8.43				
H	=	13.03				
H	=	4.45				
EARLIEST	DIST	RESS ST	ARTING TIME		0.0	YEARS
LATEST D	STRE	SS STAI	TING TIME		10.0	YEARS
DISTRESS	AT I	HITIAL	TIME	100		
EARLIEST	TIME	FRON I	. TO H		0.0	YEARS
LATEST T	INE F	ROH L	N DI	*	4.0	YEARS
EARLIEST	TIME	FROM	t TO H	=	9.0	YEARS
LATEST T	INE F	ROM M	TO H		4.0	YEARS
MUNIXAN	PRENI	CTION	AGE	=	30.0	YEARS

ALITURA ANTREZ					
INITIAL TIME		ı	0.0	YEARS	
TIME FROM L TO	M =		1	YEARS	
TIME FROM M TO	H =	:	3	YEARS	
HEAN				13.2719	YEARS
STANDARD	BEVI	ATION		3.5158	YEARS

	9.1110	Wh activities -	0.0.00	1 CANO
YEAR	L+M+H	L	M	н
•	.01	.01	0.00	0.00
1	.03	.02	.01	0.00
2	.97	.94	.03	0.00
3	.18	.11	.07	0.00
4	.42	.24	.17	.01
5	.93	.51	.39	.03
6	1.73	1.00	.86	.07
7	3.72	1.79	1.76	.18
8	6.49	2.97	3.30	.42
•	11.22	4.53	5.74	.93
10	17.60	6.38	9.29	1.93
11	25.91	8.30	13,08	3.72
12	35.89	9.97	19,22	6.69
13	46.92	11.04	24.46	11.22
14	58.20	11.28	29.32	17.60
15	68.85	10.45	32.29	25.91
16	78.11	7.26	32.97	35.88
17	85.55	7.44	31.19	46.92
19	71.04	5.51	27.35	58.20
19	94.83	3.77	22.22	48.85
20	97.21	2.38	16.73	78.11
21	78.40	1.39	11.47	85.5 \$
22	99.35	.75	7.54	91.06
23	79.72	.37	4,51	94.83
24	99.88	.17	2.50	97.21
25	79.96	.07	1.26	98.40
26	77.78	.03	.41	99.35
27	77.77	.01	.27	99.72
28	100.00	.00	.11	17.88
29	100.00	.00	.04	99.94
30	100.00	.00	.01	97.98

CRACK PREDICTION FOR ENTIRE RUNWAY WIDTH (NOTE 100% = 197050 LINEAR FEET OF CRK)

DISTRESS INPUT DATA

DISTRESS	TYPE		₹.			
AGE	-	11.00	YEARS			
L	-	4.63				
N		8.75				
M		4.57				
EARLIEST	DISTR	ESS ST	ARTING TIE	IE =	0.0	YEARS
LATEST D	ESTRES	S STAR	SHIT BHIT		9.0	YEARS
DISTRESS	AT IN	HITIAL	TIME .	0100		
EARLIEST	TIME	FROM L	. TO H		0.0	YEARS
LATEST T	INE FR	ION L T	O H	-	4.0	YEARS
EARLIEST	TIME	FROM N	TO H		0.0	YEARS
LATEST T	LHE FA	ON H T	O H		4.0	YEARS
MUNIXAN	PREDIC	NOIT:	GE	•	30.0	YEARS

	OPTINUM VALUES			
	INITIAL TIME		0.0 YEARS	
	TIME FROM L TO) H =	1 YEARS	
	TIME FROM M TO) H =	2 YEARS	
	HEAN		= 14.5310	YEARS
	STANDARD	DEVIATION	1 = 3.8474	YEARS
YEAR	L+N+N	L	H	H
0	.01	.01	0.00	0.00
1	.02	.01	.01	0.00
2	.04	.03	.02	0.00
3	.14	.08	.05	.01
4	.31	.17	.11	.02
5	.67	.35	.25	.06
6	1.34	-47	.53	.14
7	2.52	1.19	1.02	.31
	4.47	1.97	1.56	.47
•	7.54	3.05	3.15	1.34
10	11.74	4.42	5.02	2.52
11	17.95	5.77	7.47	4.49
12	25.54	7.59	10.41	7.54
13	34.54	7.00	13.58	11.76
14	44.52	9.97	16.59	17.95
15	54.85	10.33	18.97	25.54
14	64.86	10.01	20.30	34.54
17	73.93	9.07	20.35	44.52
18	81.42	7.69	17.07	54.85
17	87.71	4.07	14.76	64.86
20	9 2.23	4,51	13.78	73.93
21	75.34	3.13	10.61	81.42
22	97.38	2.03	7.64	87.71
23	78.41	1.23	5.15	72.23
24	77.30	.70	3.25	75.36
25	99.47	.37	1.92	97.38
26	99.85	.18	1.06	78.41
27	77.74	.00	.55	77.30
28	99.97	.04	.27	99.47
29	77.77	.01	.12	77.85
30	100.00	.01	-05	99.94

CRACK PREDICTION FOR OUTSIDE 75 FEET ONLY (NOTE 100% = 105325 LINEAR FEET OF CRK)

APPENDIX E: FUTURE M&R COSTS FOR EACH M&R ALTERNATIVE

ANA	LYSIS PERIODYEAR	RS INTERES	ST RA	TE
INFLATION RATE				
YEAR	M&R WORK DESCRIPTION	COST #	f	PRESENT WORTH
80	SEAL & PATCH CRACKS	51,053		
80	PATCH ALLIGATOR CRACKING	5,361		
80	APPLY REJUVENATOR	45,333		
82	SEAL & PATCH CRACKS	41,403		
84	SEAL & PATCH CRACKS	43,213		
86	SEAL & PATCH CRACKS	83,960		
87	TACK COAT	11,333		
87	OVERLAY	543,150		
87	APPLY REJ. CONST. COAT	34,000		
89	SEAL & PATCH CRACKS	10,000 *		<u> </u>
91	SEAL & PATCH CRACKS	20,000 *		
93	SEAL & PATCH CRACKS	30,000 *		
95	SEAL & PATCH CRACKS	30,000 *		
95	APPLY REJUVENATOR	45,333		
97	SEAL & PATCH CRACKS	20,000 *		
99	SEAL & PATCH CRACKS	30,000 *		
				
		TOT.	AL	5

*ESTIMATED ASSUMING REFLECTION CRACKING

M E	M & R ALTERNATIVE OVERLAY IN 1980 - KEACH PCI = 40 (ABOUT 1996) ANALYSIS PERIOD 20 YEARS INTEREST RATE 9				
ANA					
		INFLATI	ON R	ATE9	
YEAR	MAR WORK DESCRIPTION	COST \$	f	PRESENT WORTH #	
80	SEAL & PATCH CRACKS	51,053			
80	PATCH ALLIGATOR CRACKS	5,361			
80	TACK COAT	11,333		<u> </u>	
80	OVERLAY	543,150			
80	APPLY REJ. CONST. COAT	34.000			
82	SEAL E PATCH CRACKS	10,000*			
84	SEAL EPATCH CRACKS	20,000*		<u> </u>	
80	SEAL EPATCH CRACKS	30,000 *			
88	SEAL & PATCH CRACKS	30,000*			
88	APPLY REJUVENATOR				
90	SEAL & PATCH CRACKS	20,000 *			
92	SEAL & PATCH CRACKS	30,000 *			
94	SEAL É PATCH CRACKS	30,000 *			
960	SEAL & PATCH CRACKS	30,000 *			
960	APPLY REJUVENATOR				
98	SEAL & PATCH CRACKS	30,000 *			
		<u></u>			
		<u> </u>	AL	\$	
SAI I	/AGE VALUE = = :				
JAL	_				
	PRESENT WORTH = \$_				
	MATER ASSUMING BETTER				

M 8	M & R ALTERNATIVE REPLACE SURFACE COURSE (3INCH)				
ANA	ANALYSIS PERIOD 20 YEARS INTEREST RATE				
		INFLATI	ON RA	TE	
YEAR	M&R WORK DESCRIPTION	COST #	f	PRESENT WORTH #	
80	COLD MILL SURFACE	255,000			
80	HAULING MILLED MATERIAL	9,053			
80	PRIME BASE	22,667			
80	NEW BINCH AC	814,725			
80	APPLY REJ. CONST. COAT	34,000			
88	SEAL & PATCH CRACKS	1,000			
08	APPLY REJUVENATOR	45,333		<u> </u>	
92	SEAL & PATCH CRACKS	5,000			
94	SEAL & PATCH CRACKS	10,000*		 	
96	SEAL & PATCH CRACKS	15,000 *		↓	
96	APPLY REJUVENATOR	45,333			
98	SEAL & PATCH CRACKS	10,000 *		↓	
				 	
				 	
	}	 		 	
	 				
	 	 			
	 	 		 	
		_		 	
				 	
<u> </u>		+		 	
	<u> </u>	1		+	
	 	 		 	
		T01	AL	\$	
SALI	/AGE VALUE = =	<i>\$</i>			
	PRESENT WORTH = \$				

ESTIMATED ASSUMING REFLECTION CRACKING

AS BASE - ADD NEW SURFACE ANALYSIS PERIOD YEARS INTEREST RATE				
		INFLATIO	ON R	ATE
EAR	M&R WORK DESCRIPTION	COST #	f	PRESENT WORTH
80	COLD MILL	255,000		
80	WIND ROWING MILLED MATL	4,526		
80	PLACE MILLED MATL. AS			
	BASE & COMPACT & PRIME	136,000		
80	NEW AC SURFACE			
	- CENTRAL 75 FT.	407, 362		
	- OUTSIDE 75 FT.	271,575		
80	APPLY REJ. CONST. COAT	34,000		
90	SEAL & PATCH CRACKS	1,000*		
90	APPLY REJUVENATOR	45, 333		
94	SEAL & PATCH CRKS.	5,000		
98	SEAL EPATCH CRACKS	10,000 #		1
				1
				1
			 	1
				
				1
				1
		<i>T01</i>	AL	\$
	VAGE VALUE = = :	&		

,	ANAL	YSIS PERIOD ZO YEAR	RS INTERES	ST RA	TE
			INFLATIO	ON RA	1TE
YE	AR	M&R WORK DESCRIPTION	COST #	f	PRESENT WORTH
- 3	30	COLD MILL 75 FT.	127,500		
<u>_£</u>	30	HAULING MILLED MAT'L.	4,526		<u> </u>
_£	30	PRIME BASE	11,333		<u> </u>
		NEW AC SURFACE	407,362		
_£	30	SEAL & PATCH OUTSIDE CRKS			
	30_	APPLY REJ. CONST. COAT (15)			
	30	APPLY REJ. OUTSIDE	17,000		
	38	SEAL & PATCH CRACKS	500 *		
	38	APPLY REJUVENATOR	ZZ, 666		
		SEAL & PATCH CRACKS	Z500 *		
75 PT.	24_	SEAL & PATCH CRACKS	5,000 *		1
	96	SEAL & PATCH CRACKS	7.500 *		
	76	APPLY REJUVENATOR	22,666		<u> </u>
\-\frac{\frac{1}{2}}{2}	78_	SEAL & PATCH CRACKS	5,000 *		
7	32	SEAL & PATCH CRACKS	17,473		
	34	SEAL & PATCH CRACKS	21,392		1
	36	SEAL EPATCH CRACKS	39,002		1
	38	SEAL & PATCH CRACKS	31,987		1
	38	APPLY REJUVENATOR			1
	10	SEAL & PATCH CRACKS	29,449		
	12	SEAL & PATCH CRACKS	42,425		
	14	SEAL & PATCH CRACKS	33,103		
	16	SEAL & PATCH CRACKS APPLY REJUVENATOR	29,734		
	760		22,666		T _a
<u>_</u> 2	18	SEAL & PATCH CRACKS	42,478 TOT	AL	5
[]	•	AGE VALUE = = 1	•		

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